AD-AD34 169

ADA 034169





AMMRC TR 76-42

PRODUCTION ENGINEERING MEASURES PROGRAM MANUFACTURING METHODS AND TECHNOLOGY

PARALLEL AND INTERSECTING WELDS IN HIGH-STRENGTH STEEL SHEET

DONALD C. BUFFUM Army Materials and Mechanics Research Center Watertown, Massachusetts 02172

December 1976

Final Report

Approved for public release; distribution unlimited.

Prepared for

U.S. ARMY MISSILE COMMAND Huntsville, Alabama 35809

DTIC QUALITY INSPECTED 3

The findings in this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

Mention of any trade names or manufacturers in this report shall not be construed as advertising nor as an official indorsement or approval of such products or companies by the United States Government.

DISPOSITION INSTRUCTIONS

Destroy this report when it is no longer needed.

Do not return it to the originator.

REPORT DOCUMENTATION I	PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER	2. GOVT ACCESSION NO.	
AMMRC TR 76-42		
4. TITLE (and Subtitle)		5. TYPE OF REPORT & PERIOD COVERED
PARALLEL AND INTERSECTING WELDS HIGH-STRENGTH STEEL SHEET	IN	Final Report
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(#)		8. CONTRACT OR GRANT NUMBER(s)
Donald C. Buffum		
9. PERFORMING ORGANIZATION NAME AND ADDRESS		
Army Materials and Mechanics Research	anal Cart	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
Watertown, Massachusetts 02172		D/A Project: PEMA
DR XMR - ER		AMCMS Code: 2597.03.3035.00
11. CONTROLLING OFFICE NAME AND ADORESS		12. REPORT DATE
U. S. Army Missile Command		December 1976
Huntsville, Alabama 35809		13. NUMBER OF PAGES
		17
14. MONITORING AGENCY NAME & ADDRESS(if different	from Controlling Office)	15. SECURITY CLASS. (of this report)
		Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)		

Approved for public release; distribution unlimited.

17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, ff different from Raport)

18. SUPPLEMENTARY NOTES

This project has been accomplished as part of the U. S. Army Manufacturing Methods and Technology Program, which has as its objective the timely establishment of manufacturing process, techniques or equipment to insure the efficient production of current or future defense programs.

19. KEY WORDS (Continue on reverse side if necessary and identify by block number)

High-strength steels Gas tungsten arc welding Welded joints Missile components

20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

(SEE REVERSE SIDE)

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

Block No. 20

ABSTRACT

The objective of this project was to determine interaction effects of parallel and intersecting welds in high-strength steel sheet as well as to determine the need for excessive buildup of the welded area. Included are the effects of minor variables in welding upon the weldments. The alloys used in the project were 4130, 4340, and 18% Ni maraging steel sheet. The joining process was gas tungsten arc welding.

CONTENTS

	Page
INTRODUCTION	1
OBJECTIVE	1
MATERIALS AND PROCEDURES	1
RESULTS AND DISCUSSION	4
CONCLUSIONS	1·1
ACKNOWLEDGMENT	11
APPENDIX. COMPILATION OF TEST DATA	13

INTRODUCTION

In the manufacture of missile and rocket cases, the fabrication of component parts into the end item quite often requires parallel weldments in close proximity and weldments that intersect those parallel welds. Because the weld deposit is an as-cast structure and because the temperatures of welding affect the microstructure of the parent metal in the vicinity of the weld, the overall properties of a welded component can be detrimentally affected by such weldments. Normally, the deleterious effects are compensated for by additional filler metal in the weldment. This safety step increases the cost, fabrication time, material, and weight of the end item. The present program investigated the joint geometries mentioned above to determine the extent to which additional filler metal buildup was required.

This project was funded as part of the U.S. Army Missile Command Manufacturing Methods and Technology Program.

Guidance in the selection of materials, the spacing between parallel weldments, and geometry of test specimens were received from the Missile Command which sponsored the program.

OBJECTIVE

The objective of this program is to determine the minimum buildup required on adjacent and intersecting weldments and to determine the effects, if any, of the spacing between the parallel welds upon the properties of the weldments.

MATERIALS AND PROCEDURES

The materials used in this program include sheets of AISI 4130, AISI 4340, and 18% nickel maraging steel and filler wires designated as Linde 140, Airco AX-140, Linde 71, and 18% nickel maraging steel. The composition and thicknesses of the sheet material used are given in Table 1. The composition and diameters of the filler wire used are given in Table 2.

	Thick- ness						1	Weigh	t Per	cent						
Designation		С	Mn	Р	S	Ni	Cr	Мо	Cu	A1	Со	В	Zr	Ca	Ti	Si
4130	0.032	0.28	0.45	0.009	0.024	0.06	0.92	0.17	0.11	-	_	-	-	-	-	0.23
4130	.050	.30	.51	.009	.024	.06	.93	.20	.12	-	-	-	-	-	-	.34
4130	.063	.29	.54	.008	.025	.07	.94	.18	.13	-	-	-	_	_	_	.26
4340	.063	.39	.63	.009	.023	1.81	.81	.24	.14	-	-	-	-	-	-	.30
18% Ni Maraging	.030	.007	.04	.005	.006	18.15	-	4.93	-	0.10	7.32	0.003	0.02	0.06	0.43	.02
18% Ni Maraging	.055	.007	.04	.005	.006	18.15	-	4.93	-	.10	7.32	.003	.02	.06	.43	.02

Table 1. THICKNESS AND COMPOSITION OF BASE MATERIALS

The objective in determining the heat treatments was to develop a treatment that would produce a tensile strength of approximately 250,000 psi. After some initial investigations, the following heat treatments were used throughout the program.

For the 4130 and the 4340 materials in all thicknesses, the heat treatment consisted of heating to 1550 F (842 C) for 1 hour, oil quenching to room temperature, and tempering at 450 F (232 C) for 1 hour, and air cooling. Although the tensile strengths for the different thicknesses of the 4130 steel decreased with decrease in thickness of the sheet, the sponsor agreed to keep the heat treatment constant for this material.

For the 18% Ni maraging steels in both thicknesses, the heat treatment consisted of heating to 1500 F (815 C) for 1 hour, air cooling to room temperature, aging at 900 F (482 C) for 3 hours, and air cooling to room temperature.

The tensile properties produced in the various sized sheets by the above heat treatments are given in Table 3.

In preparation for welding, the sheet materials were machined into sections of appropriate size. The edges to be welded were ground parallel and square. The entire surface of the section was sand-blasted to remove oxidization and provide uniformity. This was followed by wire brushing with a clean, stainless steel wire brush. Next, the sections were washed in a biodegradable detergent solution and the faying surfaces cleaned with acetone. Just before welding, the faying surfaces were draw-filed and wiped with a clean dry cloth.

Table 2. DIAMETER AND COMPOSITION OF FILLER WIRE

,	Diameter	Weight Percent														
Designation	(in.)	С	Mn	Si	Ni	Cr	Мо	Р	S	A1	Ti	Со	٧			
Linde 140	0.045	0.11	1.65	0.35	2.90	0.70	0.90	0.008	0.007	-	-	_	-			
Airco AX-140*	0.045	0.07- 0.11	1.75- 2.0	0.25- 0.45	2.0-2.5	0.85- 1.10	0.5- 0.6	-	-	-	-	-	-			
Linde 71	0.030	0.31	0.55	0.65	-	1.25	0.5	-	-	-	-	-	0.25			
18% Ni Maraging	0.030	0.033	0.036	0.10	18.88	-	5.12	0.003	0.011	0.106	0.66	9.04	-			

^{*}Nominal analysis

Table 3. TENSILE PROPERTIES* OF HEAT-TREATED SHEET

Alloy	Thickness (in.)	Yield Strength 0.2% (ksi)	Tensile Strength (ksi)	Elongation (%)
4130	0.032	191	215	2.8
4130	.050	199	238	5.0
4130	.063	216	253	5.3
4340	.063	213	263	6.7
18% Ni	.030	239	246	4.2
18% Ni	.055	238	245	4.7

^{*}Average of three tests

The welding program was planned to include welds with no filler and welds with filler. Further, the heat treatments listed above would be given as prewelding heat treatments, post-welding heat treatments and both. The spacing between the parallel welds would be 1/2, 3/4, 1, and 1-1/4 inches. The gas tungsten arc welding process was used throughout. For backing of the weldments, a copper plate was machined with grooves 1/16" deep and 3/16" wide with appropriate separation between the grooves to match the spacers used in making the parallel welds. The welding setup is shown in Figure 1. During the early stages of the project it was necessary to change and modify the backup plate. Because the new backup plate was a less efficient heat sink it was necessary to change the welding variables to obtain satisfactory weldments. Throughout the project, it was noted that on the materials used, especially the thinner materials, relatively small changes in variables could produce a wide range of effects upon the weldments, varying from lack of penetration to burn-through. These variations will be discussed later in the report. In the initial development of settings for the welding variables, four variables were altered. These were the amperage, the voltage, the travel speed, and the filler wire rate of feed for those welds where filler wire was used. Experience showed that a wide range of settings for these variables could be used with satisfactory results. However, for simplicity and more consistent results, the travel speed and filler wire feeding rate were held constant and only the amperage and voltage were varied. Table 4 lists the welding variables finally selected. Since some variations had to be used in these settings, they should be used as a guide rather than rigid numbers to produce satisfactory weldments. It will also be noted that the travel speed for the 4340 material welded with filler wire is slower than that for the other materials. This was required because the diameter of the filler wire was greater than that used on the other materials.



Figure 1. GTA welding setup for parallel and intersecting welds
19-066-1412/AMC-73

All welds were made in the same direction using run-on tabs. When used, the filler wire was fed into the arc at an angle of 75° from the tungsten electrode. The height of the wire feed guide was adjusted so that the wire would make contact with the sheet to be welded at a distance in front of the electrode equal to the diameter of the filler wire.

RESULTS AND DISCUSSION

The experimental data were obtained by testing flat tension specimens, Figure 2, and notched flat tension specimens, Figure 3. As requested by the sponsor, to obtain results that would simulate actual production conditions, the weldments were tested in their as-welded configuration without having the reinforcement of the weld ground flat. This produced an inaccuracy in the testing of the intersecting welds. The longitudinal weld would increase the cross-sectional area by

Table 4. W	IELDING	VARIABL	ES
------------	---------	---------	----

Base Material	Thick- ness (in.)	Filler Material	Diam. (in.)	Wire Feed (in./min)	Amperage	Voltage
4130 4130 4130 4340 18% Ni 18% Ni 4130 4130 4130 4340 18% Ni	0.032 .050 .063 .063 .030 .055 .032 .050 .063 .063	None None None None None Linde 71 Linde 71 Linde 77 Airco AX-140	- - - 0.030 .030 .030 .045	- - - - - 33 33 33 33 33	70-75 100-105 130-135 125-130 70 110-120 70 96-102 120-130 114 67.5-70	10-20 11-13 11-13 8.5-11.5 10.5-12 12 11 11-12 11.5-13 13 11-13

Note:

Electrode: W-2Th, 0.062 in. diam., 30° incl. tip angle

Travel speed: 13 in./min

Argon shielding gas: 50 cfh flow

Type A B C E F G T W
TF 4 8 2.0 2-1/2 1/2R 2-1/4 3/4 0.030-0.063 0.500

For Weld Specimens
Weld Zone to be in Middle of Gage Length

W±0.002

Figure 2. Flat tension specimens

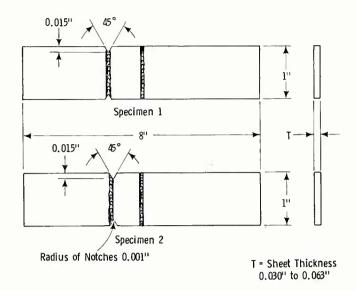


Figure 3. Notched flat tension specimen

an amount of buildup. This could not be measured, therefore the cross-sectional area was based on the width and thickness of the sheet. The results obtained will tend to be higher than they actually are. Three specimens were tested in each condition and the average of these results are listed in the Appendix, Tables A-1 to A-3.

With the large number of variables used in obtaining the data in this report, there is no clear and concise method by which the data can be presented graphically. Therefore, the data has been tabulated in the Appendix thereby enabling the reader to use it as he will.

In Table A-1, the data on single weldments without spacing are presented.

In Table A-2, the data on the parallel weldments with spacings of 1/2, 3/4, 1, and 1-1/4 inches are presented.

In Table A-3, the data on intersecting weldments are presented. In this case, a single weld intersects two parallel welds at right angles.

In Table A-4, the sharp notch tension test data are reported.

In all tables, the conditions for the specimens are listed as A, B, C, etc. These are defined as follows:

Condition	Description
A	As-received material welded without filler and then fully heat treated with the heat treatment designated in the text for that material.
В	As-received material welded with filler metal and given the appropriate heat treatment.
С	Heat-treated material welded without filler and post-weld heat treated.

D Heat-treated material welded with filler; no post-weld heat treatment.

E Heat-treated material welded with filler and post-weld heat treated.

Data are grouped according to the spacing between parallel welds. In various groups all conditions are not covered because the data obtained under certain conditions were so low as to be of no interest on this project.

Toward the end of the program, the sponsor indicated a desire to obtain some notch toughness data. It was determined that a modification of the Double-Edge-Notch Tension Specimen described in ASTM publication E-338-7 would be used, see Figure 3. Since we are interested in weld strength, the notches were originally placed on the centerline of one of the weldments (Figure 3, Specimen 1). The notch was changed to the heat-affected zone (HAZ) of one of the weldments (Figure 3, Specimen 2). The data are listed in Table A-4 of the Appendix. Because the request for notch toughness data was made late in the program and material for all conditions was no longer available, the data obtained are relatively spotty.

The notch concentration factor for the notched tension specimens is calculated with the equation $\frac{1}{2}$

 $K = \sqrt{1/2W/r}$

W = distance between notches in inches r = radius of notches in inches For the specimen used, K = 18.71.

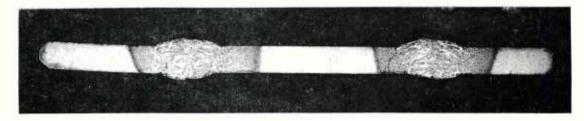
Random specimens were selected for metallographic examination. Representative photomicrographs of parallel weldments with 1/2" spacing are shown in Figure 4.

Before discussing the effects of weld spacing on the properties of the weldment, it is believed to be of value to discuss the minor variables that occurred during the performance of this project. These so-called minor variables had large effects upon the actual welding performed.

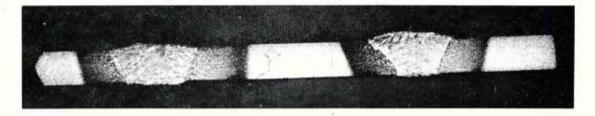
The initial work was done using an existing backup plate which was grooved for the parallel welds. While this was being used, a second backup plate was being made that could be used for both parallel and intersecting welds. When the second plate was put into the welding setup, it was observed that the welding variables determined while using the first plate no longer applied. The thickness of the two plates was not the same, therefore their function as a heat sink was different and the welding variables had to be adjusted to accommodate this difference. It was also observed that the thinnest sheet quite often cracked on welding. Although, preheat would overcome this deficiency, it was not desirable to introduce this variable into the welding process. It was thought that decreasing the heat-withdrawing property of the heat sink would correct the situation. This was accomplished by placing a one-inch-wide piece of glass tape on the backup plate 3/16" back from the centerline of the groove. This slight alteration eliminated the cracking.



4130 Steel, 0.032" Thick, Condition B



4130 Steel, 0.050" Thick, Condition B



4130 Steel, 0.063" Thick, Condition B



18% Ni Maraging Steel, 0.055" Thick, Condition D

Figure 4. Representative photomicrographs of parallel welds -1/2" spacing. Mag. 5X

On one occasion a new coil of filler wire was introduced into the welding process, followed by a deterioration of the welds produced. Microscopic examination of the filler wire surface revealed that it was contaminated to a far greater degree than the previous coil used. By replacing the contaminated coil than an uncontaminated coil of the same wire, the troubles were eliminated.

When making welds with 1/2" spacing, it was observed that at the end of the weld a high degree of mismatch and lack of penetration was produced. It was finally observed that the stress produced during welding lifted the 1/2-inch strip. By improving the hold-down fixturing for the 1/2-inch strip, this problem was corrected.

At one time a series of welds were produced in which the last third to quarter of the weldment lacked penetration. This was overcome by changing the ground from being attached to a corner of the backup plate to being attached to both ends of the backup plate at a point near to the centerline of the weld.

When intersecting welds were introduced into the program, it was noted that a lack of penetration developed in the area of the intersections. This was initially blamed on the slow response of the welding equipment to the change in height of the base metal where the buildup of the parallel welds occurred. The motor driven control was replaced with a solid state electronic device. This reduced the amount of lack of penetration but did not eliminate it. The next step was to grind off the weld bead buildup of the parallel welds at the faying surfaces to a 45° angle. This completely eliminated the lack of penetration in this area.

A series of welds were made that varied from lack of penetration to complete burn-through. After an extensive investigation, it was observed that the faying surface of the sheets being welded were contaminated. The contamination was of two types. The first was in the form of dark glassy spots that occurred intermittently along the ground edge of the plate. This was believed to be caused during the grinding of the edge in material preparation. The second contamination was in the form of very small bits of paper attached to the ground surface. These came from the paper towel used to wipe the ground edge prior to welding. By draw filing the ground surfaces and wiping this surface with a cloth, this problem was eliminated.

At times the wire feed mechanism produced trouble in the weldments. To summarize, the angle at which the wire is introduced must be kept close to 75° from vertical. The height at which it is fed or the spot at which it is introduced to the weld puddle must be constant. Also, it must be adjusted with the tip of the tungsten in the position it will be during welding. Finally, the coil of wire should be mounted in the same plane as that of arc travel to reduce its sideward motion during welding. Having the tip of the wire guide as close as possible to the nozzle of the torch will also aid in reducing the side-to-side motion of the wire.

Although it is not critical, the flow of the shielding gas must be adjusted to the conditions under which the weld is made. At first a minimum flow rate was used to reduce cost but it was found that the break in the hold-down plates, in the vicinity of the parallel welds, caused lack of protection at these points. Randomly increasing the flow rate caused irregularities due to arc blow. By making adjustments in the flow rate, both troubles were eliminated.

Inconsistent results obtained in the initial phases of the project were traced to variations in the geometry of the tip of the tungsten electrode. The literature* indicates the extent to which electrode geometry can affect the welding operation. By grinding the vertex angle of the electrode to 30° and maintaining this angle, the inconsistencies were drastically reduced.

^{*}SAVAGE, W. F., STRUNEK, S. S., and ISHIKAWA, Y. The Effect of Electrode Geometry in Gas Tungsten-Arc Welding. Welding Journal, v. 44, November 1965, p. 489s-496s.

On the thin sheet weldments that were heat treated prior to welding, the lack of flatness in the sheet plus the distortion developed by the heat and stress of welding caused the sheet to buckle slightly and separate from the backup plate, thereby reducing the heat flow characteristics. This would result in burn-through. By improving the hold-down clamping device, this problem was eliminated.

Once the voltage, current, travel speed, and wire feed speed were established, relatively small changes in these variables had no appreciable effect upon the weldments produced.

One of the studies made on the project was the path which the fracture took in the breaking of the flat tension specimens. The following determinations were made for parallel weldments.

- 1. Weldments made with filler metal and post-weld heat treatment broke predominantly in the base metal.
- 2. Weldments welded without filler metal and with or without post-weld heat treatment broke either in the heat-affected zone or in the weld.
- 3. Weldments made with filler metal and without post-weld heat treatment broke about half in the parent metal and half in the HAZ.

For those weldments with parallel and intersecting welds:

- 1. Weldments made with filler and with post-weld heat treatment again broke predominantly in the parent metal.
- 2. Weldments made without filler but with post-weld heat treatment broke in the HAZ.

In reviewing the mechanical test data, it must be taken into consideration that the strengths reported are based on the cross-sectional area of the base metal not of the weld area. As a result, those values for specimens that broke through the weld may have an actual strength lower than that reported. For the intersecting welds, all actual values should be lower than those reported. No practical method could be devised in which the actual cross-sectional area could be measured accurately. It should be further noted that the cross-sectional area will vary with the amount of buildup on the weldment.

In general, all weldments without filler broke at strengths lower than that of the base material. As a result, these weldments were sometimes omitted in the parallel weld series and all were omitted from the intersecting weld series.

All weldments welded with filler but without post-weld heat treatment gave tensile values quite a bit below that of the base metal.

Weldments made with filler and given a post-weld heat treatment had values close to that of the parent metal with half being above and half below. All but four had tensile strength above a value equal to 90% of the tensile strength of the parent metal.

In all tests, these values did not vary with the different materials used.

It is further noted that those specimens with two parallel weldments generally have lower tensile strengths than the comparable specimens with a single weldment. This may be related to the possibility that when making the second weldment the material being welded did not seat as well on the backup plate. This would be caused by buildup on the underside of the first weldment. Any void between the material being welded and the copper backup plate would cause a variation in heat flow which could be detrimental to the weld being made. Unfortunately, no records were taken that would indicate the sequence in which the parallel welds were made.

Since the path for fracture for those specimens that broke in the weldment was predominantly through the heat-affected zone, the amount of buildup in the weldment is not critical. In other words, no matter how much buildup is used, the weakest area of the weldment will be at the edge of the built-up zone. To add an excess of filler metal will only move the weakened area a little further from the centerline of the weld. It was also noted that for the welding conditions used in this project, the heat-affected zone was relatively narrow and would not affect the adjacent weld.

In the case of intersecting welds, the main fault to avoid is lack of penetration in the area of intersection. Although weld buildup at the intersection could be used to compensate for this weakness in the flaw, this would increase costs by requiring more filler metal and labor to provide the buildup. A sound welding procedure that would prevent the development of the flaw would be the better solution. Also, attempts to repair the lack of penetration flaws by manual repair welding produced more problems rather than correcting the situation.

To obtain an indication of the notch toughness of the weldments, it was decided to make sharp notch tension tests and compare the data obtained with that for standard tension specimens. Originally, two notches were placed at the centerline of one of the welds, one notch at each end of the weld. A study of the path of fracture showed that the fracture started in the weld at the notch and usually propagated to the heat-affected zone. This meant that the cross-sectional area varied for each specimen and accurate notched specimen tensile strengths could not be determined. Later, the notches were placed with their centers in the heat-affected zone of the weldment and more reproducible results were obtained.

Because of the items noted above, it is not possible to draw definite conclusions from the sharp notch tension test data of the welded specimens. However, it can be concluded that the sharp notch tensile strength of the material was not drastically affected by welding.

Selected parallel weldments were examined metallographically to determine whether or not the heat-affected zones of the weldments overlapped. Figure 4 shows representative photomicrographs. It will be noted that as the thickness of the parent metal increased, the width of the weld-affected area increased. Although no photomicrographs of the 4340 material are shown, they were comparable to those of the 4130. It will also be noted that the width of the weld-affected

area for the 18% Ni maraging steel was greater than that for the other materials. However, in no case was there an overlap of the two areas. This again would indicate that there were no interrelated effects of the parallel welds.

CONCLUSIONS

- 1. In welding high-strength steel alloys in relatively thin cross sections, it is mandatory that welding procedures be determined and maintained for producing sound welds.
- 2. Excessive buildup of weldments will not necessarily improve the strength of the joint.
- 3. Minor variations in electrode configuration, wire feeding geometry, joint preparation and fixturing of the joint to be welded can have a large effect on the soundness of the welded joint.
- 4. Manual repair of lack of penetration faults in intersecting welds is not a feasible solution.
- 5. Minor changes in welding variables of voltage, amperage, travel speed, and wire feeding rate did not affect the mechanical properties of the weldment.
- 6. There are no interrelated effects between parallel welds down to 1/2-inch spacing between weldments.

ACKNOWLEDGMENT

The author wishes to commend Mr. Daniel NcNaught for his patience, diligence, and keen observation of anomalies in the weldments developed during welding. Without his full cooperation, many of the so-called minor problems would have been overlooked and the successful completion of the project would have been severely hampered.

APPENDIX. COMPILATION OF TEST DATA

Condition Description As-received material welded without filler and then fully Α heat treated with the heat treatment designated for that matrial. В As-received material welded with filler metal and given the appropriate heat treatment. C Heat-treated material welded without filler and post-weld heat treated. D Heat-treated material welded with filler; no post-weld heat treatment. Ε Heat-treated material welded with filler and post-weld heat treated.

Table A-1. TENSION TEST DATA ON SHEETS WITH SINGLE WELDMENTS

5pacing	Condi- tion	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon. (%)	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.
5ingle Weld	A 8 C 0 E	206 210 209 196 208	- 0.03 244 257 242 197 245	2.7 3.4 1.8 0.5 2.8	4130 201 198 197 181 199	- 0.05 244 239 238 181 240	0" T 4.0 4.8 5.0 1.0 5.0	222	0 - 0.06 cracked 258 omitted 217 268	3.8	239 207 242	cracked 281 omitted 209 287	5.0	18% N 231 223 236 151 243	i - 0.0 234 227 238 153 250	1.2 2.7 2.0 2.0 3.0	18% N 234 241 242 155 253	1 - 0.0 236 245 243 159 257	2.0 4.5 2.8 2.5 3.5

Table A-2. TENSION TEST OATA ON SHEETS WITH PARALLEL WELOMENTS

5pacing	Condi- tion	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.	Y.5. 0.2% (ksi)	T.5. (ksi)	Elon.
1/2 in.	A 8 C	4130 148 198 161 171	- 0.03 169 228 185 174	2.3 2.0 1.7	4130 179 210 189	210 251 216	3.7 3.8 2.7	215 184	- 0.06 omitted 251 218	5.0 5.7	205 244 206	- 0.06 242 291 255	3.7 3.7 1.5	214 238 222	218 242 230	1.7 2.0 0.8	220 235 217	228 241 226	4.4 2.8 2.5
3/4 in.	A 8 C	153 204 149 113	236 182 240 172 127	0.7 3.0 2.2 1.8 3.3 1.8	158 183 - 216 170 167	168 218 - 254 190 178	1.0 4.5 4.3 2.8 2.5	157 197 216 185 159	166 231 - 252 213 172	2.5 5.0 - 4.8 4.2 2.0	214 226 214 183	252 263 256 193	1.5 4.8 3.5 4.0 4.2 1.0	214 226 214 226 233 131	220 241 238 146	2.0 1.8 2.7 2.0 2.5 1.8	211 231 215 138	214 237 229 146	2.5 2.8 3.0 3.2 2.8 3.2
l in.	A 8 C E	157 - 177 139 197	213 152 227	3.5 4.2 0.3 2.7	183 - 186 191 183	216 223 232 217	3.9 3.8 4.2 3.5	211 203 193 203	255 237 224 237	3.8 4.7 1.6 4.5	217 215 211 216	257 226 255 255	4.7 2.5 3.8 2.8	234 - 234 205 251	242 240 211 257	3.2 3.2 3.2 1.2	227 239 220 251	235 - 248 225 258	3.5 - 3.8 3.7 2.7
1-1/4 in.	8 C E	172 160 166	227 168 191	3.8 1.5 3.0	190 184 198	222 205 234	3.2 2.7 4.8	203 193 207	240 231 255	5.5 1.1 2.8	223 215 216	263 256 263	3.0 3.8 3.7	238 208 233	247 213 244	3.3 2.5 2.5	233 212 243	241 220 249	3.0 5.3 2.7

Table A-3. TENSION TEST DATA ON SHEETS WITH PARALLEL AND INTERSECTING WELDMENTS

5pacing	Condi- tion	Y.5. 0.2% T.5. Elon. (ksi) (ksi) (%)	Y.5. 0.2% T.5. Elon. (ksi) (ksi) (%)	Y.5. 0.2% T.5. Elon. (ksi) (ksi) (%)	Y.5. 0.2% T.5. Elon. (ksi) (ksi) (%)	Y.S. 0.2% T.5. Elon. (ksi) (ksi) (%)	Y.5. 0.2% T.5. Elon. (ksi) (ksi) (%)
1/2 in.	8 C E	4130 - 0.032" T 190 210 1.3 cracked 203 210 1.3	4130 - 0.050" T 210 226 2.3 cracked 210 226 3.0	4130 - 0.063" T 206 231 2.3 174 183 2.2 175 182 1.2	4340 - 0.063" T 121	18% Ni - 0.030" T 251 257 2.0 219 229 2.0 245 250 1.7	18% N1 - 0.055" T 247 251 2.0 231 242 3.3 251 257 3.0
3/4 in.	8	187 207 1.2	omitted	196 206 2.0	100 149 2.5	251 255 1.8	250 255 3.0
	C	cracked	180 194 2.2	199 204 2.0	- 176 1.2	237 239 3.0	211 223 1.5
	E	burn-through	184 202 3.0	209 227 2.5	237 273 4.3	222 229 3.0	252 256 2.3
l in.	8	192 206 1.8	206 238 1.0	208 225 3.7	232 283 4.9	247 258 3.0	234 243 4.2
	C	168 178 1.7	190 218 2.7	198 207 1.7	230 255 2.0	217 226 3.0	220 229 4.2
	E	191 197 2.0	199 215 2.7	199 225 3.3	226 275 5.0	262 275 2.0	253 264 3.0
1-1/4 in.	8	192 202 0.7	200 221 2.3	210 228 1.6	227 274 5.0	254 260 2.3	234 243 4.8
	E	192 202 4.0	206 219 1.8	194 205 1.5	229 277 4.8	250 259 3.2	231 240 4.3

Table A-4. SHARP NOTCH TENSION TEST OATA*

							labie	N-4. 31	. SHARP NOTCH TENSION TEST GATA*										
5pacing	Condi- tion	Std. T.5. (ksi)	Notch T.5.† (ksi)		5td. T.5. (ksi)	Notch T.5.† (ksi)		5td. T.5. (ksi)	Notch T.5.† (ksi)		5td. T.5. (ksi)	Notch T.5.† (ksi)		5td. T.5. (ksi)	Notch T.5.† (ksi)		5td. T.5. (ksi)	Notch T.5.† (ksi)	
		4130	- 0.0	32" T	4130	- 0.0	50" T	4130	- 0.0	63" T	4340	- 0.0	63" T	18%	li - 0.	030" T	18%	ii - 0.	055" T
5ingle Weld	А	244	171 171	0.70 0.70	242	134 139	0.55 0.57	-	-	-	_	-	-	234	182 198	0.78	236	222 226	0.94
:	В	257	159 215	0.62 0.84	239	146 195	0.61 0.82	258	134 134	0.52	281	181 220	0.64	-	-	-	-	-	-
	С	242	150 150	0.62 0.62	238	154 154	0.65 0.65	-	-	-	-	-	-	-	-	-	-	-	-
	0	-	-	-	-	-	-	217	72 95	0.33	-	-	-	-	-	-	-	-	-
	Ε	245	159 206	0.65 0.84	240	164 183	0.68 0.76	268	118 154	0.44	287	188 217	0.66 0.76	227	167 255	0.74 1.12	245	196 241	0.80
	8ase Material	215	209	0.97	253	148	0.58	238	192	0.81	263	171	0.65	245	243	0.99	246	251	1.02
1/2 in.	8	228	150 275	0.66 1.21	251	147 217	0.59 0.86	251	153 207	0.61 0.82	-	-	-	242	161 294	0.67	241	171 277	0.71
	С	-	-	-	212	187 187	0.88 0.88	-	-	-	-	-	-	-	-	-	226	217 237	0.81
	0	174	65 90	0.37 0.52	168	100 162	0.60	-	-	-	-	-	-	148	100 178	0.68	146	123 179	0.69
	E	236	135 247	0.57 1.05	223	140 227	0.63 1.02	231	157 242	0.68 1.05	-	-	-	233	150 172	0.64	232	192 284	0.68
3/4 fn.	8	240	147 246	0.61 1.03	254	160 246	0.63 0.97	252	168	0.67 0.93	263	164 263	0.62 0.98	241	162 283	0.67	237	186 267	0.78
	С	174	161 163	0.93 0.94	190	111 111	0.58 0.58	-	-	-	-	-	-	238	171 302	0.72 1.27	229	185 217	0.81
	0	127	106 189	0.83 1.49	178	96 146	0.54 0.82	-	-	-	-	-	-	146	102 171	0.70 1.17	146	120 148	0.82
	E	182	86 195	0.47 1.07	216	149 245	0.69 1.13	228	166 222	0.73 0.97	257	137 252	0.54 0.99	235	201 283	0.86 1.20	235	201 283	0.86
1 in.	8	213	202	0.95	233	197	0.88	255	202	0.79	226	177	0.78	240	246	1.03	248	243	0.99
	E	-	-	-	-	-	-	-	-	-	-	-	-	257	205	0.80	251	237	0.94
1-1/4	8	227	200	0.88	222	205	0.92	240	189	0.79	-	-	_	247	244	0.99	241	236	0.98
in.	Ε	-	-	-	-	-		_	_	-	_	_	-	244	209	0.86	249	227	0.90

*Notches in weld (spacings - single weld, 1/2 in., and 3/4 in.)
Notches in heat-affected zone (spacings - 1 in. and 1-1/4 in.)
†Top value calculated on cross-sectional area at weld, bottom on area of sheet

No. of Copies Tο Copies Tο Director, U. S. Army Materiel Systems Analysis Activity, Aberdeen Proving Ground, Maryland 21DD5 Office of the Director, Defense Research and Engineering, The Pentagon, Washington, D. C. 2D3D1 12 Commander, Defense Documentation Center, Cameron Station, Building 5, 501D Duke Street, Alexandria, Virginia 22314 Commander, U. S. Army Electronics Command, 225 South 18th Street, Philadelphia, Pennsylvania 19103
1 ATTN: DRSEL-PP/P/IED-2, Mr. Wesley Karg Advanced Research Projects Agency, The Pentagon, Washington, D. C. 2D315 Commander, U. S. Army Mobility Equipment Research and Development Command, Fort Belvoir, Virginia 22D6D Metals and Ceramics Information Center, Battelle Columbus Laboratories, 5D5 King Avenue, Columbus, Dhio 432D1 ATTN: Mr. Daniel Maykuth 2 ATTN: Technical Documents Center, Building 315 Commander, U. S. Army Production Equipment Agency, Manufacturing Technology Brancn, Rock Island Arsenal, Illinois 612D2 Chief of Research and Development, Department of the Army, Washington, D. C. 2D310 ATTN: Physical and Engineering Sciences Division Dr. Bernard R. Stein ATTN: DRXPE, Mr. Ralph Siegel Commander, U. S. Army Research and Engineering Directorate, Warren, Michigan 48D90 ATTN: SMOTA-RCM.1, Mr. Edward Moritz SMOTA-RCM.1, Mr. Donald Phelps Commander, U. S. Army Materiel Development and Readiness Command, 5D01 Eisenhower Avenue, Alexandria, Virginia 22333 ATTN: DRCDE-DE, Development Division DRCDE-RS, Research Division Commander, Watervliet Arsenal, Watervliet, New York 12189 DRCDE-RS, Scientific Deputy ATTN: SARWY-R DRCDE-TC 1 Dr. Robert Weigle Commander, U. S. Army Aviation Systems Command, P. D. Box 209, Main Office, St. Louis, Missouri 63166 ATTN: DRSAV-LEP, Mr. J. M. Thorp Chief, Bureau of Naval Weapons, Department of the Navy, Room 2225, Munitions Building, Washington, D. C. DRSAV-ER, Dr. I. Peterson Chief, Bureau of Ships, Department of the Navy, Washington, D. C. 2D315
1 ATTN: Code 341 Commander, U. S. Army Missile Command, Redstone Arsenal, Alabama 358D9
ATTN: DRSMI-IE, Mr. J. E. Kirshtein
DRSMI-R, Mr. John L. McDaniel
DRSMI-RBLD, Redstone Scientific Information Center Chief of Naval Research, Arlington, Virginia 22217 1 ATTN: Code 471 Chief Scientist, Dr. W. W. Carter Naval Research Laboratory, Washington, D. C. 20375 2 ATTN: Dr. G. R. Yoder - Code 6382 Directorate of R&D 1 Dr. B. Steverding Commander, U. S. Army Mobility Equipment Research and Development Command, 43DO Goodfellow Boulevard, St. Louis Missouri 6312D Hq, USAF/RDPI, The Pentagon, Washington, D. C. 20330 1 ATTN: Major Donald Sponberg Headquarters, Aeronautical Systems Division, 4950 TEST W/TZHM (DH 2-5 Mgr), Wright-Patterson Air Force Base, Ohio 45433 ATTN: AFML/MATB/Mr. George Glenn ATTN: DRSME-PLC, Mr. J. Murphy Commander, U. S. Army Armament Command, Rock Island, Illinois 61201 AFML/MXE/E. Morrissey AFML/LLP/D. M. Forney, Jr. ATTN: DRSAR-SC, Dr. C. M. Hudson DRSAR-PPW-PB, Mr. Francis X. Walter AFML/MBC/Mr. Stanley Schulman 2 Technical Library National Aeronautics and Space Administration, Commander, U. S. Army Tank-Automotive Research and Development Command, Warren, Michigan 48090 ATTN: DRDTA-PPS, Mr. David Siegel Washington, D. C. 2D546 ATTN: AFSS-AD, Dffice of Scientific and Technical Information Mr. B. G. Achhammer Mr. G. C. Deutsch, Chief, Materials Research Program, Mr. J. P. Jones Code RR-1 Commander, Aberdeen Proving Ground, Maryland 21DD5 ATTN: Technical Library, Building 313 National Aeronautics and Space Administration, Lewis Research Center, 21000 Brookpark Road, Cleveland, Ohio 44135 1 ATTN: Mr. G. Mervin Ault, Assistant Chief, M&S Division Commander, U. S. Army Foreign Science and Technology Center, 220 7th Street, N. E., Charlottesville, Virginia 229DI ATTN: DRXST-SD2 National Aeronautics and Space Administration, Marshall Space Flight Center, Huntsville, Alabama 35812 ATTN: S&E-ME-MM, Mr. W. A. Wilson, Building 4720 R-P&VE-M, R. J. Schwinghamer Commander, Frankford Arsenal, Philadelphia, Pennsylvania 19137 ATTN: Pitman-Dunn Institute of Research SARFA-L3DD, Mr. J. Corrie Albany Metallurgy Research Center, Albany, Oregon 97321 1 ATTN: Mr. A. H. Roberson, Research Director Commander, Picatinny Arsenal, Dover, New Jersey D7BD1 ATTN: Feltman Research Laboratories Defense Materials Service, General Services Administration, Washington, O. C. 2D4D5 Commander, Rock Island Arsenal, Rock Island, Illinois 61201 ATTN: SARRI-RDL 1 ATTN: Mr. Clarence A. Fredell, Director, Technical R&D Staff Director, Eustis Directorate, U. S. Army Air Mobility General Dynamics, Convair Aerospace Division, P. O. Box 748, Fort Worth, Texas 761D1 ATTN: Mfg. Engineering Technical Library Research and Development Laboratory, Fort Eustis, Virginia 23604 ATTN: Mr. J. Robinson, SAVDL-EU-SS Director, Army Materials and Mechanics Research Center, Watertown, Massachusetts D2172 ATTN: DRXMR-PL Commander, U. S. Army Ballistic Research Laboratories, Aberdeen Proving Ground, Maryland 21DD5 1 ATTN: Dr. D. Eichelberger DRXMR-AG

Author

AD UNCLASSIFIED UNLIMITED DISTRIBUTION Key Words High-strength steels Gas tungsten arc welding Weided joints	nteraction effects of parallel and as well as to determine the need ded are the effects of minor vari- used in the project were 4130, ing process was gas tungsten arc	
Army Materials and Mechanics Research Center, Watertown, Massachusetts D2172 PARALLEL AND INTERSECTING WELOS IN HIGH- STRENGTH STEEL SHEET - Donald C. Buffum Technical Report AWMRC TR 76-42, December 1976, 17 pp, illus-tables, 0/A Project PEMA, AMCMS Code 2597.03.3035.0D	The objective of this project was to determine interaction effects of parallel and intersecting welds in high-strength steel sheet as well as to determine the need for excessive buildup of the welded area. Included are the effects of minor variables in welding upon the weldments. The alloys used in the project were 4130, 4340, and 18% Ni maraging steel sheet. The joining process was gas tungsten arc welding.	
UNCLASSIFIED UNLIMITED DISTRIBUTION Key Words High-strength steels Gas tungsten arc welding		
Army Materials and Mechanics Research Center, Watertown, Massachusetts 02172 PARALLEL AND INTERSECTING WELDS IN HIGH- STRENGTH STEEL SHEET - Donald C. Buffum Technical Report AMMRC TR 76-42, Oecember 1976, 17 pp, illus-tables, D/A Project PEMA, AMCMS Code 2597.03.3035.00	The objective of this project was to determine interaction effects of parallel and intersecting welds in high-strength steel sheet as well as to determine the need for excessive buildup of the welded area. Included are the effects of minor variables in welding upon the weldments. The alloys used in the project were 4130, 4340, and 18% Ni maraging steel sheet. The joining process was gas tungsten arc welding.	
Army Materials Matertown Matertown PARALLEL STRENGTH Technical Repoi	The objective (intersecting we for excessive lables in weldin 4340, and 18% welding.	